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Adaptive Eulerian framework for boiling and evaporation

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Abstract

We propose in this work an adaptive Eulerian framework for the simulation of both boiling and evaporation phenomena occurring at the interface of a heated 3D solid immersed in a liquid tank. It simultaneously takes into account the gas-liquid phase changes, the vapor formation and their dynamics, and consequently the 3D quenching or cooling of a heated solid. It uses a Level Set method to separate and to track each phase. The phase change is performed using the balance of heat fluxes at the interface without the use of conforming mesh. Instead, the use of an a posteriori error estimate leading to highly stretched anisotropic elements at the interface enables to drastically reduce errors on computed jumps. This avoids the need of interface reconstruction or interpolation procedure. Finally, a Variational Multiscale solver for the Navier-Stokes equations is extended with implicit treatment of the surface tension. A series of 2D and 3D problems are solved to highlight the efficiency and the accuracy of the proposed framework. The cooling of an immersed solid is also presented and shows good agreement with experimental data. To the best of our knowledge, direct numerical simulations of quenching using an Eulerian framework with boiling and evaporation have never been considered.

Keywords: Phase change, Multiphase flows, Immersed method, Level Set, Anisotropic mesh adaptation, Stefan problem

1. INTRODUCTION

2 An accurate and robust simulation of boiling phenomena is still an ongoing challenge. Indeed, the
3 complexity of boiling lies in the wide range of scales to consider and in the physics involved. From the
4 nucleation of vapor bubbles to film boiling and bath hydrodynamics, one needs to consider scales from
5 μm to m . Indeed, in industrial processes, such as quenching of a solid hot metallic part, the variety of
6 configurations and the complexity of the surrounding flow must also be taken into account as they have
7 direct impacts on gradients of mechanical properties, microstructure and residual stresses.

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8 The classical well known methods for thermal treatment during the cooling of a solid generally refer
9 to the use of experimentally deduced heat transfer coefficients. However, these latter are only useful for
10 a particular configuration in term of geometry, orientation of the surface, range of temperature and flow
11 motion. Therefore, heat transfer coefficients are not suitable to analyze different industrial processes and
12 cannot be generalized to all configurations. Only a direct numerical simulation can take into account all the
13 physics and all the scales involved in boiling.

14 An abundant research has been devoted to the modeling of the boiling and the evaporation phenomena.
15 The approaches to simulate these phenomena can be summarized into two categories that depend on the
16 thickness of the interface: sharp interface and diffuse interface. The sharp interface approach considers
17 the presence of different phases that interact through the interface using boundary conditions. Within this
18 approach, two methods are commonly used. First, in interface tracking methods, the interface is represented
19 by a set of nodes. This set of nodes moves accordingly to the interface. This requires particular mesh move-
20 ments at each iteration and also requires a special treatment of complex topological configurations [1–5].
21 Alternatively, interface capturing methods consist in a implicit definition of the interface. A volume fraction
22 (Volume-Of-Fluid) or a signed distance function (Level Set) is convected using a transport equation [6–18].
23 These methods are popular due to the simplicity of implementation and the fact that complex topology
24 changes such as the coalescence of bubbles are handled naturally. A literature review of these methods
25 applied to boiling is given by Kharangate and Mudawar in [19]. Other approaches in the literature can
26 be mentioned. Phase-field approaches seem promising [20–22]. An order parameter is used to distinguish
27 different phases and its evolution gives access to the interface location. The interface is assumed to have a
28 thickness of a few nanometers. However, an equation of state is required for the fluid which introduces com-
29 plexity from a mathematical and numerical point of view. Furthermore, this approach introduces high-order
30 differential terms that may be stiff and require innovative discretization techniques. We can also mention
31 the fluid mixtures approaches [23, 24] considering two compressible phases, where a complete hyperbolic
32 system is derived using conservation equations for each phase. An equation of state that reproduces the
33 phase diagram is usually required to close the system.

34 We propose in this work an adaptive Eulerian framework for the simulation of both boiling and evapora-
35 tion phenomena occurring at the interface of a heated 3D solid immersed in a liquid tank. It was shown that
36 an adaptive Eulerian framework is very efficient for the simulation multiphase flows [25–28], compressible
37 and incompressible flows [29] and yield stress fluids [30]. This framework takes into account the gas-liquid
38 phase changes, the vapor formation and their dynamics, and enables consequently the simulation of 3D
39 quenching or cooling of a heated solid. A level set method is used to separate and track the different phases.
40 The phase change is performed using the balance of heat fluxes at the interface without the use of conform-
41 ing mesh. Instead, the use of an a posteriori error estimate [31–33] leading to highly stretched anisotropic
42 elements at the interface enables to drastically reduce errors on computed jumps. This avoids the need

43 of interface reconstruction or interpolation procedure. We derive the Navier-Stokes equations taking into
 44 account the mass and energy transfer between phases at the interface. A series of 2D and 3D problems are
 45 solved to highlight the efficiency and the accuracy of the proposed framework. The cooling of an immersed
 46 solid is also presented and shows good agreement with experimental data. To the best of our knowledge,
 47 direct numerical simulations of quenching using an Eulerian framework with boiling and evaporation have
 48 never been considered.

49 The paper is structured as follows. In Section 2, we present the governing equations for the fluid motion
 50 and the level set transport equation. In Section 3, we recall the main steps of the anisotropic mesh adaptation
 51 procedure governed by the length distribution tensor. In Section 4, we provide some numerical results and
 52 examples to assess the capability of the proposed method.

53 2. Eulerian framework

54 This section is devoted to the presentation of the Eulerian framework. First, the interface capturing
 55 method is presented. It enables to follow the evolution of the interface between phases and to distribute in
 56 space the respective physical properties by defining a mixing law. From this mixing law, a phase change
 57 model will be derived and the Navier-Stokes equations will be consequently modified to take into account
 58 the phase change model.

59 2.1. Level Set approach

60 In this section, we describe the level set method used to locate the interface between the liquid phase
 61 and the vapor phase. It is a signed distance function and it is a widely used tool in different fields such as
 62 crystal growth, image restoration or surface reconstruction [34]. Let Ω be the whole domain, Ω_l the liquid
 63 domain and Ω_v the vapor domain. The level set function is a signed distance function from the interface
 64 $\Gamma = \Omega_l \cap \Omega_v$ defined at each node X of Ω as follows:

$$\alpha(X) = \begin{cases} -\text{dist}(X, \Gamma) & \text{if } X \in \Omega_l, \\ 0 & \text{if } X \in \Gamma, \\ \text{dist}(X, \Gamma) & \text{if } X \in \Omega_v. \end{cases} \quad (1)$$

65 The evolution of the level set function is described by the following transport equation [35]

$$\frac{\partial \alpha}{\partial t} + u \cdot \nabla \alpha = 0, \quad (2)$$

66 where u is a velocity. The level set, as a distance function, verifies $\|\nabla \alpha\| = 1$. However, when the interface
 67 is convected by u , the **level set** can lose this property, which may cause numerical instabilities, and needs
 68 to be reinitialized to recover it. A common way to reinitialize it, is to solve the following Hamilton-Jacobi
 69 equation [35]

$$\frac{\partial \alpha}{\partial \tau} + s(\alpha)(\|\nabla \alpha\| - 1) = 0, \quad (3)$$

70 where τ is a pseudo-time and $s(\alpha)$ is the sign function of α . The steady state solution of this non-linear
71 hyperbolic equation will be a distance function from the interface while keeping unchanged its zero isovalue.

72 Once the level set function is computed, we can distribute in space the corresponding physical properties
73 (for example ρ_v and ρ_l , respectively vapor and liquid densities) using a mixing law as follows:

$$\rho = H(\alpha)\rho_v + (1 - H(\alpha))\rho_l, \quad (4)$$

74 where H is a smoothed Heaviside function used to obtain a better continuity at the interface and given by

$$H(\alpha) = \begin{cases} 1 & \text{if } \alpha > \varepsilon, \\ \frac{1}{2} \left(1 + \frac{\alpha}{\varepsilon} + \frac{1}{\pi} \sin \left(\frac{\pi \alpha}{\varepsilon} \right) \right) & \text{if } |\alpha| \leq \varepsilon, \\ 0 & \text{if } \alpha < -\varepsilon. \end{cases} \quad (5)$$

75 Here ε is a small parameter such that $\varepsilon = O(h_{\text{im}})$, known as the interface thickness, and h_{im} is the mesh
76 size in the normal direction to the interface. In the vicinity of the interface, it can be computed using

$$h_{\text{im}} = \max_{j,l \in K} \nabla \alpha \cdot \mathbf{x}^{jl}, \quad (6)$$

77 where K is the mesh element under consideration and $\mathbf{x}^{jl} = \mathbf{x}^l - \mathbf{x}^j$ represents the edge connecting the
78 nodes \mathbf{x}^j and \mathbf{x}^l of K .

79 2.2. Phase change model

80 In this section, we derive a pseudo-compressible model accounting for mass transfer at the interface.
81 From the mass conservation equation, we will quantify the exchange of mass between the vapor and the
82 liquid phases. To take into account this exchange of mass, the Navier-Stokes equations and the level set
83 equation will be modified. Such derivation has already be done in the previous work of Denis [36], in the
84 context of finite difference method.

85 2.2.1. Derivation of the governing equations for the phase change

86 We recall the Navier-Stokes equations:

$$87 \quad \rho(\partial_t u + u \cdot \nabla u) - \nabla \cdot (2\mu \varepsilon(u)) + \nabla p = f_{ST} + f, \quad (7)$$

$$88 \quad \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho u) = 0, \quad (8)$$

89 where u , p , ρ , μ , f_{ST} and f are the velocity, the pressure, the density, the viscosity, the surface tension force
 90 and additional source term of the momentum equation respectively.

91 The derivation of the model relies on the mixing law used to distribute the density in space. Therefore,
 92 we first define a mixing law for the density

$$\rho = (\rho_v - \rho_l)H(\alpha) + \rho_l, \quad (9)$$

93 where H is a Heaviside function. The mass conservation in the domain Ω reads

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho u) = 0. \quad (10)$$

94 We define a surface mass transfer rate $\dot{m}[kg.m^{-2}.s^{-1}]$ to quantify the exchange of mass between the liquid
 95 and vapor phases:

$$\frac{\partial}{\partial t} (\rho_v H(\alpha)) + \nabla \cdot (\rho_v H(\alpha) u) = \dot{m} |\nabla \alpha| \delta(\alpha), \quad (11)$$

$$\frac{\partial}{\partial t} ((1 - H(\alpha)) \rho_l) + \nabla \cdot (\rho_l (1 - H(\alpha)) u) = -\dot{m} |\nabla \alpha| \delta(\alpha), \quad (12)$$

98 where δ is a Dirac function. Expanding (11) and (12) leads to

$$\rho_v \frac{\partial H(\alpha)}{\partial t} + \rho_v H(\alpha) \nabla \cdot u + \rho_v u \cdot \nabla H(\alpha) = \dot{m} |\nabla \alpha| \delta(\alpha), \quad (13)$$

$$-\rho_l \frac{\partial H(\alpha)}{\partial t} + \rho_l (1 - H(\alpha)) \nabla \cdot u - \rho_l u \cdot \nabla H(\alpha) = -\dot{m} |\nabla \alpha| \delta(\alpha). \quad (14)$$

101 By dividing (13) and (14) by their respective density and summing, we obtain the new mass conservation
 102 equation

$$\nabla \cdot u = \dot{m} \left(\frac{1}{\rho_v} - \frac{1}{\rho_l} \right) |\nabla \alpha| \delta(\alpha). \quad (15)$$

103 The velocity is not divergence free since a mass transfer occurs at the interface between the vapor and the
 104 liquid.

105 Since a transfer of mass occurs, the level set equation will be modified to advect the level set not only
 106 from the velocity obtained through the resolution of the Navier-Stokes equations but also the velocity of the
 107 vapor front. This is obtained by summing (13) and (14) and dividing by $(\rho_v - \rho_l)$:

$$\frac{\partial H(\alpha)}{\partial t} + u \cdot \nabla H(\alpha) = \frac{\rho}{\rho_l - \rho_v} \nabla \cdot u. \quad (16)$$

108 Considering the derivative in time and in space of the Heaviside function, $\frac{\partial H(\alpha)}{\partial t} = \frac{\partial H(\alpha)}{\partial \alpha} \frac{\partial \alpha}{\partial t} = \delta(\alpha) \frac{\partial \alpha}{\partial t}$ and
 109 $\nabla H(\alpha) = \delta(\alpha) \nabla \alpha$, we obtain

$$\delta(\alpha) \frac{\partial \alpha}{\partial t} + \delta(\alpha) u \cdot \nabla \alpha = \frac{\rho}{\rho_l - \rho_v} \nabla \cdot u. \quad (17)$$

110 In the case of a divergence free velocity fields, Eq (17) resumes to the standard level set equation. Replacing
 111 $\nabla \cdot u$ by (15) leads to

$$\delta(\alpha) \frac{\partial \alpha}{\partial t} + \delta(\alpha) u \cdot \nabla \alpha = \frac{\rho}{\rho_l - \rho_v} \dot{m} \left(\frac{1}{\rho_v} - \frac{1}{\rho_l} \right) |\nabla \alpha| \delta(\alpha). \quad (18)$$

112 Extending to the whole domain and simplifying leads to

$$\frac{\partial \alpha}{\partial t} + u \cdot \nabla \alpha = \frac{\rho}{\rho_l \rho_v} \dot{m} |\nabla \alpha|. \quad (19)$$

113 The level set equation now reads

$$\frac{\partial \alpha}{\partial t} + \left[u - \frac{\rho}{\rho_l \rho_v} \dot{m} \frac{\nabla \alpha}{|\nabla \alpha|} \right] \cdot \nabla \alpha = 0. \quad (20)$$

114 The level set is now advected by the velocity u obtained from the resolution of the Navier-Stokes and by
 115 a velocity representing the vaporization of the water.

116 Similarly, one can use the same derivation for the energy equation. Neglecting the heat generated by
 117 viscosity forces and capillary forces, the energy equation reads

$$\rho c_p \left(\frac{\partial T}{\partial t} + u \cdot \nabla T \right) - \nabla \cdot (k \nabla T) = - (L + (c_p^v - c_p^l)(T - T_{\text{sat}})) \dot{m} \delta(\alpha) |\nabla \alpha| \frac{\rho^2}{\rho_v \rho_l}. \quad (21)$$

118 where T is the temperature, T_{sat} is the saturation temperature, c_p is the specific heat, c_p^v (resp. c_p^l) is the
 119 specific heat in the vapor (resp. in the liquid) and k is the thermal conductivity. The formulation accounts
 120 naturally for the thickness of the interface. It is a generic formulation that requires the definition of the
 121 mass transfer rate \dot{m} .

122 *Remark 1.* The smoothing of the sharp mass flux presented in [37], using an averaged density is automatically
 123 obtained in (19). The origin of the smoothing is due to the smoothing of the distribution of the density
 124 across the interface.

125 2.2.2. Derivation of the surface mass transfer rate

126 The mass transfer rate is resolved using the heat released by an elementary volume dV of liquid that has
 127 vaporized between the time t and $t + dt$. We consider a domain composed of water and vapor (see Figure 1).
 128 The position of the interface at the time t is given by $\alpha(t)$, dS and n being respectively the corresponding
 129 elementary surface and its normal pointing in the vapor direction.

130 The heat released dQ by a volume dV of liquid that has vaporized during the elapsed time between t
 131 and $t + dt$ is:

$$dQ = \rho_l L dV \quad (22)$$

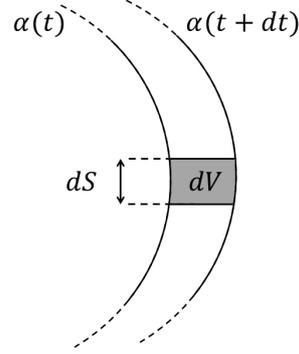


Figure 1: Volume dV that has vaporized between t and $t + dt$.

132 where L is the enthalpy of vaporization [$J.kg^{-1}$]. The heat fluxes are defined respectively in the vapor and
 133 the liquid by $\phi_v = -k_v \nabla T_v$ and $\phi_l = -k_l \nabla T_l$ where k_v and k_l are the thermal conductivity in the vapor
 134 and the liquid respectively. Vaporization (resp. condensation) occurs when the jump of fluxes across the
 135 interface $[[\phi]] = [\phi_v - \phi_l]_{|\alpha=0} \cdot n$ is positive (resp. negative). Then dQ reads:

$$dQ = \int_{t'=t}^{t+dt} [[\phi]] dS dt'. \quad (23)$$

136 Differentiating and taking the limit for $dt \rightarrow 0$ leads to

$$\lim_{dt \rightarrow 0} \rho_l L \frac{\alpha(t + dt) - \alpha(t)}{dt} = \lim_{dt \rightarrow 0} \frac{1}{dt} \int_{t'=t}^{t+dt} (-k_v \nabla T_v + k_l \nabla T_l)_{|\alpha=0} \cdot n dt'. \quad (24)$$

137 We obtain the surface mass transfer rate, the so-called Stefan condition,

$$\dot{m} = \rho_l L \frac{d\alpha}{dt} = (-k_v \nabla T_v + k_l \nabla T_l)_{|\alpha=0} \cdot n. \quad (25)$$

138 Eq. (25) requires the evaluation of the balance of fluxes at the interface $(-k_v \nabla T_v + k_l \nabla T_l)_{|\alpha=0} \cdot n$. This
 139 evaluation is not straightforward when using an implicit definition of the interface. We therefore choose
 140 to use a delta formulation. Approximating the surface integral by mean of a delta Dirac function on each
 141 elementary volume Ω_i containing the surface S_i leads to

$$\int_{S_i} (-k_v \nabla T_v + k_l \nabla T_l)_{|\alpha=0} \cdot n dS_i = \int_{\Omega_i} \delta(\alpha) (-k_v \nabla T_v + k_l \nabla T_l) \cdot n d\Omega_i. \quad (26)$$

142 Then, integrating (25) over Ω_i gives an expression for the surface mass transfer \dot{m} :

$$\dot{m} = \frac{\int_{\Omega_i} \delta(\alpha) (-k_v \nabla T_v + k_l \nabla T_l) \cdot n d\Omega_i}{\int_{\Omega_i} \delta(\alpha) d\Omega_i}. \quad (27)$$

143 *2.3. Implicit surface tension*

144 A common way to introduce the surface tension in the Navier-Stokes equations is to use the Continuum
 145 Surface Force model [38]. It consists first in rewriting the surface force as a volume one as follows:

$$f_{ST} = -\gamma\kappa\delta\mathbf{n}. \quad (28)$$

146 where γ is the surface tension coefficient, δ is a Dirac function locating the interface, κ is the mean curvature
 147 and n is the normal to the interface. Then (28) is used as a source term in the momentum equation (8).
 148 However, an explicit implementation imposes the following severe restriction on the time step [38]:

$$\Delta t < (\Delta x)^{\frac{3}{2}} \sqrt{\frac{\rho_l + \rho_v}{4\pi\gamma}}. \quad (29)$$

149 If this time step restriction is not respected, spurious oscillations pollute the solution and destabilize the
 150 interface. However, this time step restriction penalizes the computational cost of numerical simulations since
 151 it is one or two orders of magnitude lower than the time step restriction of a CFL condition. To alleviate this
 152 restriction, we use the approach developed in [39] and widely used in [25, 40, 41]. Using differential geometry,
 153 the surface Laplacian $\Delta_s I_\Gamma$ of an identity mapping function is $-\kappa\mathbf{n}$. We apply the surface Laplacian operator
 154 Δ_S on $I_\Gamma^{n+1} = I_\Gamma^n + \mathbf{u}^{n+1}\Delta t$ which represents the evolution in time of an interface. The surface Laplacian can
 155 be decomposed into a standard Laplacian $\nabla^2 u - \frac{\partial^2 u}{\partial \mathbf{n}^2} - \kappa \frac{\partial u}{\partial \mathbf{n}}$. Multiplying by the surface tension coefficient
 156 leads to the new expression for the surface tension:

$$f_{ST} = -\gamma\kappa\delta\mathbf{n} - \gamma\delta\Delta t \left(\frac{\partial^2 \mathbf{u}}{\partial \mathbf{n}^2} + \kappa \frac{\partial \mathbf{u}}{\partial \mathbf{n}} - \nabla^2 \mathbf{u}^{n+1} \right). \quad (30)$$

157 The usual term $-\gamma\kappa\delta\mathbf{n}$ is now completed by additional terms proportional to the time step. These additional
 158 terms act as an isotropic diffusion minus a diffusion in the normal direction of the interface. More details
 159 are provided in [25].

160 *2.4. Variational MultiScale method for the Navier-Stokes equations*

161 Following the developments from previous sections, we generalize the Navier-Stokes equations, into:

162 *Find the velocity u and the pressure p such that:*

$$\rho(\partial_t u + u \cdot \nabla u) - \nabla \cdot (2\mu\varepsilon(u)) + \nabla p = f_{ST} + f, \quad (31)$$

$$\nabla \cdot u = f_c, \quad (32)$$

165 where ρ , μ , f_{ST} , f and f_c are the density, the viscosity, the surface tension force, additional source term of
 166 the momentum equation and the source term of the continuity equation respectively.

167 The stability of the discrete formulation of Navier-Stokes depends on appropriate compatibility restric-
 168 tions on the choice of the finite element spaces for the velocity and the pressure. Standard Galerkin mixed

169 elements with continuous equal order linear/linear interpolation is not a stable discretization and exhibits un-
 170 controllable oscillations that pollute the solution. The Variational MultiScale method, proposed by Hughes
 171 [42, 43], offers a general framework that deals with different and new variant of mixed variational formula-
 172 tions [44–47]. We briefly recall here the main steps.

173 First, let us consider a decomposition of the velocity and the pressure fields into resolvable coarse-scale
 174 and unresolved fine-scale:

$$175 \quad u = u_h + \tilde{u}, \quad (33)$$

$$176 \quad p = p_h + \tilde{p}, \quad (34)$$

177 Likewise, we apply the same decomposition for the weighting functions. The unresolved fine-scales are
 178 usually modeled using residual based terms that are derived consistently. The static condensation consists
 179 in substituting the fine-scale solution into the large-scale problem providing additional terms, tuned by
 180 a local stabilizing parameter. The latter enhances the stability and accuracy of the standard Galerkin
 181 formulation.

182 Second, to derive the stabilized formulation, we solve the fine scale problem, defined on the sum of
 183 element interiors and written in terms of the time-dependent large-scale variables. Then we substitute in
 184 the coarse problem, the fine-scale solution approximated within each element by:

$$185 \quad \tilde{u} = \sum_{\tau_h} \tau_u \tilde{P}_u(R_u), \quad (35)$$

$$186 \quad \tilde{p} = \sum_{\tau_c} \tau_c \tilde{P}_c(R_c), \quad (36)$$

187 where R_u and R_c are the finite element residuals. \tilde{P}_u and \tilde{P}_p are projection operators taken as the identity
 188 in this work. τ_u and τ_c are the so called stabilization parameters. Thus, we eliminate the explicit appearance
 189 of the fine-scale while still modeling their effects. More details are provided in [25, 29].

190 To simplify the notation, we use f_m and f_c as the source terms in (31)-(32), adding the explicit terms
 191 of (30) into f_m . Inserting the expression for the subscales, we finally obtain the stabilized finite element
 192 problem. The new variational formulation reads for the coarse scale,

$$\left\{ \begin{array}{l} (\rho \partial_t (u_h + \tilde{u}), v_h) + (\rho (u_h + \tilde{u}) \cdot \nabla (u_h + \tilde{u}), v_h) - (p_h + \tilde{p}, \nabla \cdot v_h) \\ + (2\mu \varepsilon(u_h) : \varepsilon(v_h)) + (\gamma \delta \Delta t \nabla (u_h + \tilde{u}) : \nabla v_h) = (f_m, v_h) \quad \forall v_h \in V_{h,0}, \\ (\nabla \cdot (u_h + \tilde{u}), q_h) = (f_c, q_h) \quad \forall q_h \in Q_h, \end{array} \right. \quad (37)$$

193 and for the fine scale,

$$\left\{ \begin{array}{l} (\rho \partial_t (u_h + \tilde{u}), \tilde{v}) + (\rho (u_h + \tilde{u}) \cdot \nabla (u_h + \tilde{u}), \tilde{v}) - (p_h + \tilde{p}, \nabla \cdot \tilde{v}) \\ + (2\mu \varepsilon(\tilde{u}) : \varepsilon(\tilde{v})) + (\gamma \delta \Delta t \nabla (u_h + \tilde{u}) : \nabla \tilde{v}) = (f_m, \tilde{v}) \quad \forall \tilde{v} \in \tilde{V}, \\ (\nabla \cdot (u_h + \tilde{u}), \tilde{q}) = (f_c, \tilde{q}) \quad \forall \tilde{q} \in \tilde{Q}. \end{array} \right. \quad (38)$$

194 At this level, two assumptions can be made to simplify the resolution of the fine scale equation: the subscales
 195 are considered quasi-static and the convection is approximated by $(u_h + \tilde{u}) \cdot \nabla (u_h + \tilde{u}) \approx u_h^c \cdot \nabla (u_h + \tilde{u})$.
 196 Therefore, by formulating the expression of \tilde{u} and \tilde{p} by substituting them into the large-scales equation, and
 197 applying integration by parts, the system to solve becomes finally

$$\left\{ \begin{array}{l} (\rho \partial_t u_h, v_h) + (\rho u_h^c \cdot \nabla u_h, v_h) - (p_h, \nabla \cdot v_h) + (2\mu \varepsilon(u_h) : \varepsilon(v_h)) + (\gamma \delta \Delta t \nabla u_h : \nabla v_h) \\ - \sum_{K \in \mathcal{T}_h} (\tau_u R_u, \rho u_h^c \cdot \nabla v_h) - \sum_{K \in \mathcal{T}_h} (\tau_c R_c, \nabla \cdot v_h) = (f_m, v_h) \quad \forall v_h \in V_{h,0}, \\ (\nabla \cdot u_h, q_h) - \sum_{K \in \mathcal{T}_h} (\tau_u R_u, \nabla q_h) = (f_c, q_h) \quad \forall q_h \in Q_h, \end{array} \right. \quad (39)$$

198 where R_u and R_c are the residuals defined by

$$\begin{aligned} R_u &= f_m - \rho \partial_t u_h - \rho u_h^c \cdot \nabla u_h - \nabla p_h, \\ R_c &= f_c - \nabla \cdot u_h. \end{aligned} \quad (40)$$

199 Note that in the case of strongly anisotropic meshes with highly stretched elements, the definition of the
 200 stabilization parameters is still an open problem and plays a critical role in the design of the stabilizing coef-
 201 ficients. In [25] the authors propose a particular choice of the stabilizing parameters suitable for anisotropic
 202 partitions that we adopt here.

203 By comparing the standard Galerkin method with the proposed stable formulation, additional integrals
 204 that are evaluated element-wise are involved. These additional terms, obtained by replacing the approxi-
 205 mated \tilde{u} and \tilde{p} into the large-scale equation, represent the effects of the sub-grid scales and above all take
 206 into account the modified surface tension terms. They are introduced in a consistent way to the Galerkin for-
 207 mulation and enable to overcome the instability of the standard formulation arising in convection dominated
 208 flows and to deal with the pressure instabilities [48].

209 3. Edge-based mesh adaptation for multiphase flows

210 Anisotropic mesh adaptation on unstructured meshes plays an important role in the efficient numerical
 211 simulation of multiphase flow problems since it helps keeping high accuracy while reducing the dedicated
 212 CPU time to these simulations [49, 50]. As it is shown in [25] and more recently in [51], anisotropic mesh
 213 adaptation outperforms adaptive octree. Small features are captured more easily using anisotropic mesh

214 than using octree. The use of anisotropic mesh adaptation enables to reduce the number of integration points
 215 by one order of magnitude [51]. The idea is to concentrate elements in regions where the solution exhibits
 216 a rapid variation of either the solution or its derivatives. In this case, it would be the large temperature
 217 gradients, the large jumps at the interfaces as well as the change in direction for the velocity fields. The
 218 objective is then to get the smallest error possible while controlling the number of mesh nodes. A vector or
 219 a scalar monitor function is used in order to control the size, shape and orientation of the elements of the
 220 mesh to be generated. This function is usually designed to give an estimate of some measure of the solution
 221 error which is then equidistributed over each mesh cell. The mesh adaptation algorithm is built in order to
 222 compute a mesh and a numerical solution. At each stage, we compute a numerical solution on the current
 223 mesh and we evaluate an estimation of the interpolation error. We set up a minimization problem that aims
 224 at minimizing the interpolation error in the L_1 -norm, independently of the problem at hand [32]. To take
 225 into account the solution development, we derive an optimal metric that minimizes its interpolation error.
 226 Therefore, a new mesh is generated according to this metric field. The originality in the approach lies also
 227 in its simplicity to compute the metric and the associated edge-based error estimator developed below.

228 3.1. Edge-based error estimation

229 Let u_h be a P1 finite element approximation obtained by applying the Lagrange interpolation operator
 230 to a regular function $u \in C^2(\Omega)$. At each vertex i of the mesh, we have $U_i = u(x^i) = u_h(x^i)$ (where x^i are
 231 the coordinates of the vertex i). Let $\Gamma(i)$ be the "patch" associated to a vertex x_i of the mesh defined as
 232 the set of nodes which share one edge with x_i , and let us denote by x_{ij} the edge connecting x_i to x_j as in
 Figure 2.

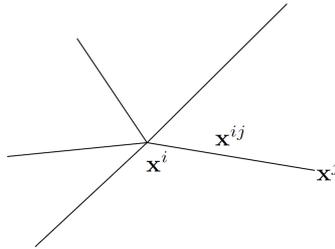


Figure 2: Patch associated with node x^i

233

234 The gradient $\nabla u^h \cdot x^{ij}$ on the edge x_{ij} is continuous, therefore we can write

$$235 \quad U^j = U^i + \nabla u^h \cdot x^{ij}, \quad (41)$$

236 which leads to

$$237 \quad \nabla u_h \cdot x^{ij} = U^j - U^i. \quad (42)$$

238 The work performed in [32] allows us to write the following error estimator:

$$239 \quad \|\nabla u^h \cdot x^{ij} - \nabla u(x^i) \cdot x^{ij}\| \leq \max_{y \in |x^i, x^j|} |x^{ij} \cdot H_u(y) \cdot x^{ij}|, \quad (43)$$

240 with H_u the Hessian of u . At the node x^i , we seek the recovered gradient g^i of u_h

$$241 \quad \nabla g_h \cdot x^{ij} = g^j - g^i. \quad (44)$$

242 We want the projection of the Hessian based on the gradient at the extremities of the edge, thus we have

$$243 \quad (\nabla g_h \cdot x^{ij}) \cdot x^{ij} = (g^j - g^i) \cdot x^{ij}, \quad (45)$$

$$244 \quad (H_u \cdot x^{ij}) \cdot x^{ij} = g^{ij} \cdot x^{ij}, \quad (46)$$

245 with $g^{ij} = g^j - g^i$. It can be shown [32] that the quantity $|g^{ij} \cdot x^{ij}|$ gives a second order accurate
 246 approximation of the second derivative of u along the edge x^{ij} . Motivated by the fact that, for P1 finite
 247 elements on anisotropic meshes, edge residuals dominate a posteriori errors, as proved in [52], it is therefore
 248 suitable to define an error indicator function associated to the edge x^{ij} as

$$249 \quad e^{ij} = |g^{ij} \cdot x^{ij}|. \quad (47)$$

250 And this error, is the exact interpolation error along the edge and allows to evaluate the global L_1 error.
 251 However, the gradient is not know at the vertices, thus a recovery procedure has to be considered.

252 3.2. Gradient recovery procedure

253 The gradient recovery procedure relies on the following optimization problem:

$$254 \quad G^i = \arg \min_G \left(\sum_{j \in \Gamma(i)} |(G - \nabla u_h) \cdot x^{ij}|^2 \right), \quad (48)$$

255 where G^i is the recovered gradient. Denoting by \otimes the tensor product between two vectors, let us introduce
 256 X^i the length distribution tensor at node i

$$257 \quad X^i = \frac{1}{|\Gamma(i)|} \left(\sum_{j \in \Gamma(i)} x^{ij} \otimes x^{ij} \right), \quad (49)$$

258 whose purpose is to give an average representation of the distribution of edges in the patch. Let us express
 259 the recovered gradient G^i in terms of the length distribution tensor

$$260 \quad G^i = (X^i)^{-1} \sum_{j \in \Gamma(i)} U^{ij} x^{ij}. \quad (50)$$

261 Therefore, the estimated error e_{ij} is thus written as

$$262 \quad e_{ij} = G^{ij} \cdot x^{ij}. \quad (51)$$

263 *3.3. Metric construction*

264 In order to relate the error indicator defined in (51) to a metric suitable for a mesh adaptation procedure,
 265 we introduce the concept of stretching factor s^{ij} defined as the ratio between the length of the edges x^{ij}
 266 after the adaptation procedure and after the adaptation procedure, [32]. The metric takes the following
 267 expression:

$$268 \quad \widetilde{M}^i = (\widetilde{X}^i)^{-1}, \quad (52)$$

269 where \widetilde{X}^i is defined as:

$$270 \quad \widetilde{X}^i = \frac{1}{|\Gamma(i)|} \left(\sum_{j \in \Gamma(i)} s^{ij} \otimes s^{ij} \right). \quad (53)$$

271 The stretching factor s^{ij} of the edge ij is chosen so that the total number of nodes in the mesh is kept fixed
 272 and is defined as

$$273 \quad s^{ij} = \left(\frac{e_{ij}}{e(N)} \right), \quad (54)$$

274 where $e(N)$ is the total error .

275 *3.4. Mesh adaption criteria*

276 In multiphase applications, the material interface between liquid and gas need to be modeled accurately.
 277 Two strategies are commonly used; we refer to them in this work as explicit and implicit adaptation. In
 278 the first one, we design and pre-adapt the mesh around the boundaries and in some regions of interest. The
 279 obtained adapted mesh will be used all along the simulation, provided that the flow exhibits a bounded level
 280 of unsteadiness. The criteria for the mesh adaptation are geometric and do not depend on the solution.
 281 Whereas, the implicit strategy imposes a dynamic mesh adaptation that changes the mesh frequently and
 282 minimizes as possible the prescribed error. Consequently, it requires a criterion based solely on the solution.

283 The common way to adapt a mesh to several variables, such as the velocity and the level set function, is
 284 to compute the metrics corresponding to each of them and then to produce a unique metric by an operation
 285 known as the intersection of metrics. In this work, we simplify this operation and we use one metric that
 286 account for different variables. Therefore, based on the theory proposed in the previous section, it is possible
 287 to extend definition (47) to account for several sources of error. In the following numerical experiments,
 288 the adaptivity accounts for the velocity components V_k , its magnitude $|V|$, the level set function α and the
 289 temperature T , by defining the following vector of sources of error:

$$\mathbf{v}(\mathbf{x}^i) = \left\{ \frac{V_k^i}{|V^i|}, \frac{|V^i|}{\max_j |V^j|}, \frac{\alpha^i}{\max(\alpha)}, \frac{T^i}{\max(T)} \right\}. \quad (55)$$

290 Because all fields are normalized (the velocity components V_x , V_y and V_z by the local velocity norm,
 291 the velocity norm $|V|$, the temperature and the level set function α by their respective global maximum), a
 292 field that is much larger in absolute value does not dominate the error estimator, and the variations of all
 293 variables are fairly taken into account.

294 3.5. High-fidelity anisotropic meshing

295 We aim to show the flexibility of the proposed mesh adaptation technique to deal with multiphase flows.
 296 Therefore we consider three fixed objects defined by level set functions inside a squared domain (see Figure 3).
 297 The circle of radius 0.1m is centered at (0.15;0.15). The square of 0.20m size is centered at (0.85;0.15). The
 298 regular pentagram is centered at (0.5;0.75) and the radius of the circumcircle is 0.2m. We choose to position
 299 them close to the wall to assess the capacity of the method to capture the features of the geometry close to
 300 a boundary.

301 In multiphase simulations, we use the level set function to define the properties in each phase. The
 302 physical properties are usually discontinuous across the interface. To avoid discontinuities which lead to
 303 numerical errors, we use a smooth Heaviside function computed from the level set function. This creates
 304 an interface transition with a thickness of few elements. The use of mesh adaptation techniques enables to
 305 reduce this thickness. As it is depicted in Figure 3, for a given number of elements (10,000), the transition
 is finer with an anisotropic adaptive mesh.

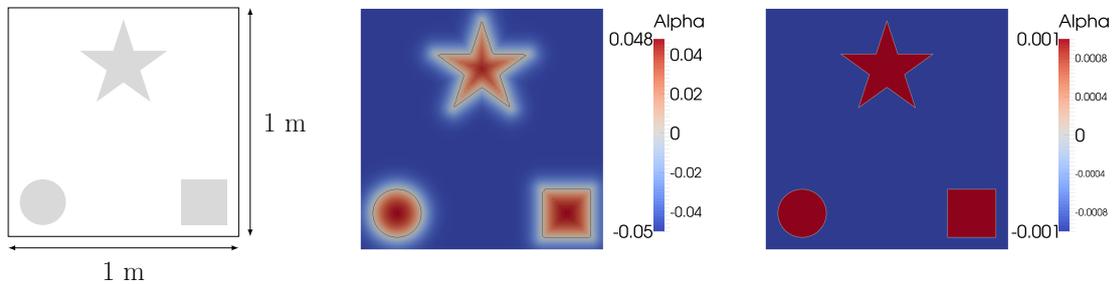


Figure 3: Three immersed objects inside a squared cavity (left). Filtered level set function for 10,000 elements in a structured mesh (middle) and in an adaptive mesh (right). The use of an adaptive mesh enables a finer interface transition. Respective meshes are shown in Figure 5.

306
 307 Figure 4 shows the obtained zero isovalue of the level set functions using different number of nodes. The
 308 comparison with structured meshes using the same number of nodes shows that anisotropic mesh adaptation
 309 allows easily to keep very good accuracy of the geometry, even for a low number of nodes. Figure 5 shows
 310 the correct orientation and deformation of the mesh elements (longest edges parallel to the boundary).
 311 This yields a great reduction of the number of triangles. These results give confidence that the proposed
 312 framework allows to deal with different shapes, with angles, singular point and curvatures.

313 Finally, in Figure 6, we measure the accuracy of the mesh adaptation technique. We compute the total
 314 perimeter and the total area of the three immersed objects and we plot the error between the analytical
 315 and the numerical solutions. We also plot the error for a structured mesh. Figure 6 confirms the advantage
 316 of using anisotropic adaptive meshes for multiphase flows. For a given accuracy, at least ten times more
 elements are required in a structured mesh.

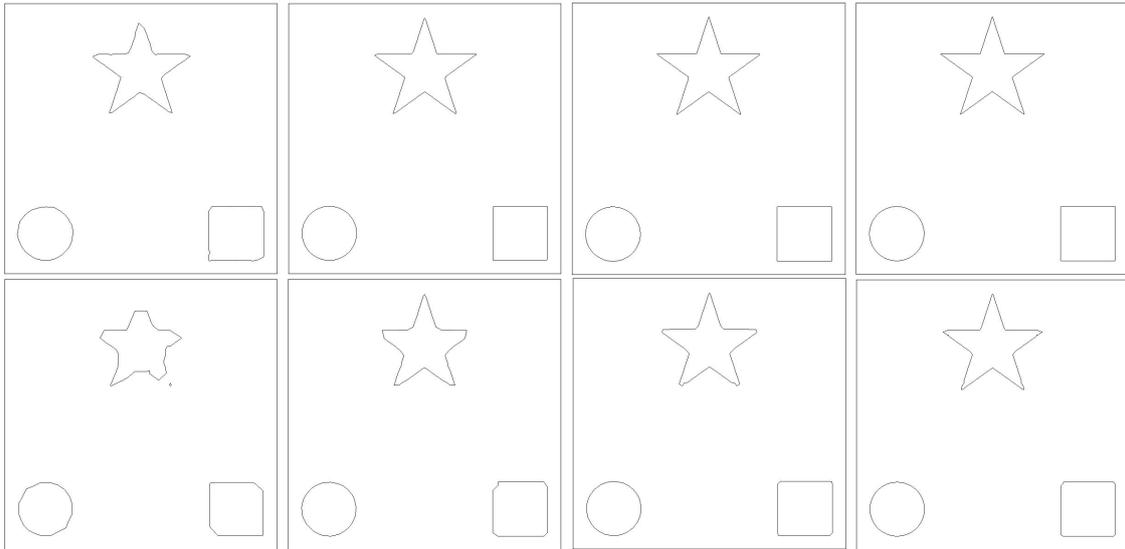


Figure 4: Zero isovalue of the level set function for 1,000, 2,000, 5,000 and 10,000 elements. First line: result with adaptive meshing. Second line: result with structured meshes.

317

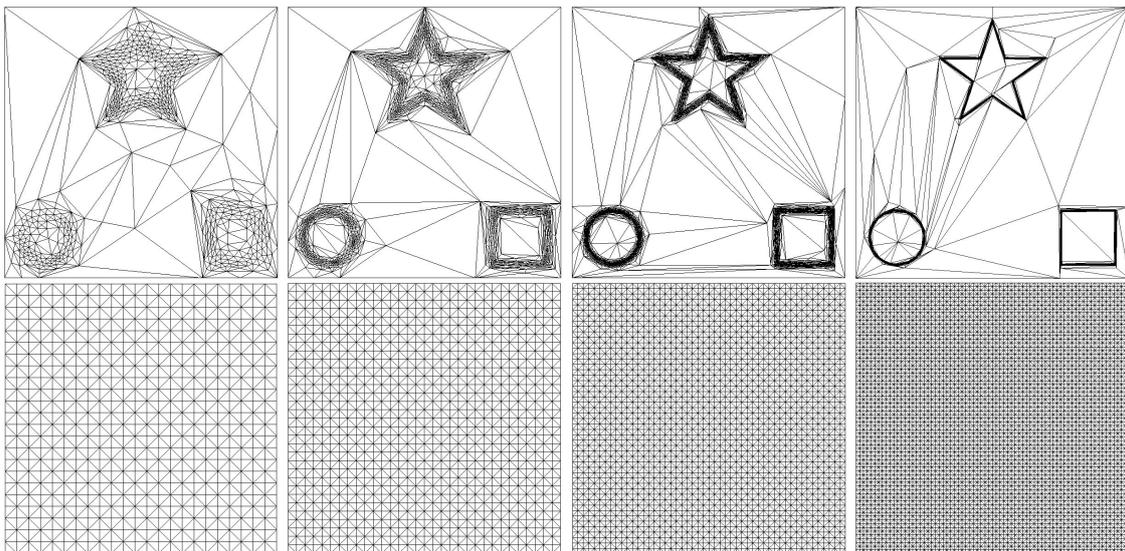


Figure 5: The obtained mesh for 1,000, 2,000, 5,000 and 10,000 elements.

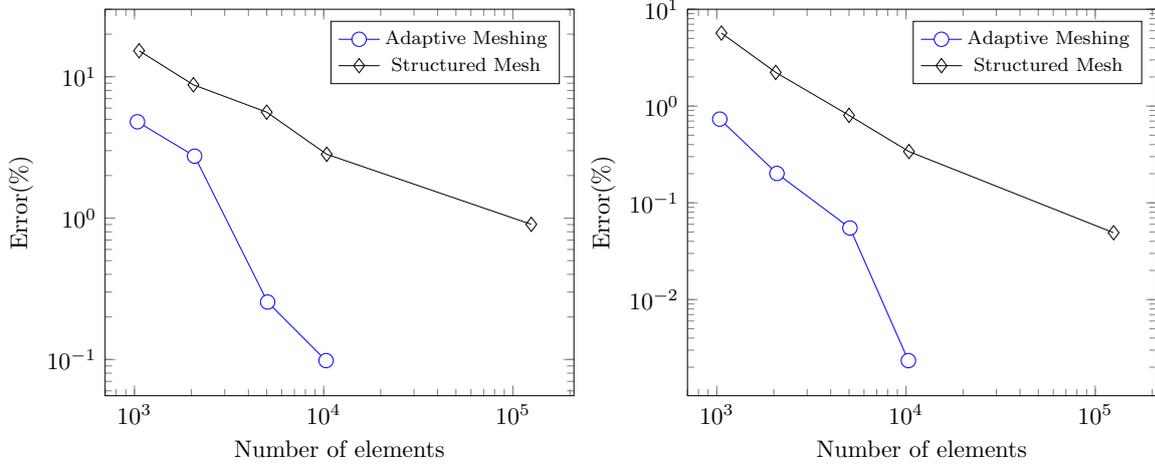


Figure 6: Percentage of error for the computation of the perimeter (left) and the area (right).

318 4. Numerical examples

319 The proposed adaptive Eulerian framework for the simulation of both boiling and evaporation phenomena
 320 is verified and validated in this section through a variety of test problems. Throughout this section, all the
 321 obtained results will be compared to either analytical or numerical available solutions. In the last test case,
 322 we bring forward a new problem with the experimental data on the cooling of a 3D heated solid inside a
 323 water tank.

324 4.1. Stefan problem

325 First, we consider the one-dimensional Stefan benchmark. It is a well known problem and serves as a
 326 benchmark to assess the accuracy of phase change models [8, 11, 53]. It is defined schematically in Figure 7.
 327 The domain is initially filled with water. The wall temperature is set constant and larger than the saturation
 328 temperature. The water is at saturation temperature. At $t > 0$, a phase change occurs and induces a motion
 329 of the interface between the vapor and the water. The convective term in the energy conservation equation
 330 is neglected in both phases.

331 The position of the interface is given by

$$s(t) = 2\chi\sqrt{\alpha_v t}, \quad (56)$$

332 where α_v is the thermal diffusivity defined by $\alpha = k_v/(\rho_v c_p^v)$ and χ is the solution of the transcendental
 333 equation

$$\frac{T_{\text{sat}} - T_{\text{wall}}}{\sqrt{\pi}L} c_p^v = \chi \operatorname{erf}(\chi) \exp(\chi^2), \quad (57)$$

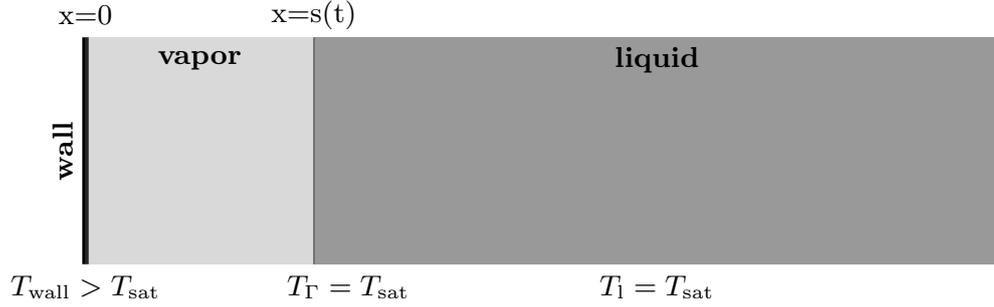


Figure 7: Initial setup for the classic Stefan problem.

334 derived in the case of a constant temperature in the liquid. The temperature in the vapor at a given time t
 335 is given by

$$T(x, t) = T_{\text{wall}} + \frac{T_{\text{sat}} - T_{\text{wall}}}{\text{erf}(\chi)} \text{erf}\left(\frac{x}{2\sqrt{\alpha_v t}}\right), \quad (58)$$

336 We consider the physical properties for the water and the vapor given in Table 1 and we consider $T_{\text{wall}} - T_{\text{sat}} =$
 337 10K . We solve the transcendental equation using a Newton algorithm to find the value of χ .

Table 1: Density, dynamic viscosity, specific heat and thermal conductivity for the vapor and the water at atmospheric pressure

	ρ [kg/m ³]	μ [Pa·s]	c_p [J/(kg·K)]	k [W/(m·K)]	L_{sat} [J/kg]
Vapor	0.597	1.26×10^{-5}	2030	0.025	
Water	958.4	2.8×10^{-4}	4216	0.679	2.26×10^6

337
 338 Figure 8 shows the evolution of the position of the interface for both the analytical and the numerical
 339 solution having good agreement. Figure 9 shows that the use of a delta Dirac function to compute the mass
 340 transfer rate results in a better convergence. This comparison was done by prescribing a maximum number
 341 of iterations (3000) to reach a residual (10^{-7}) using the GMRES method for the resolution of the linear
 342 system.

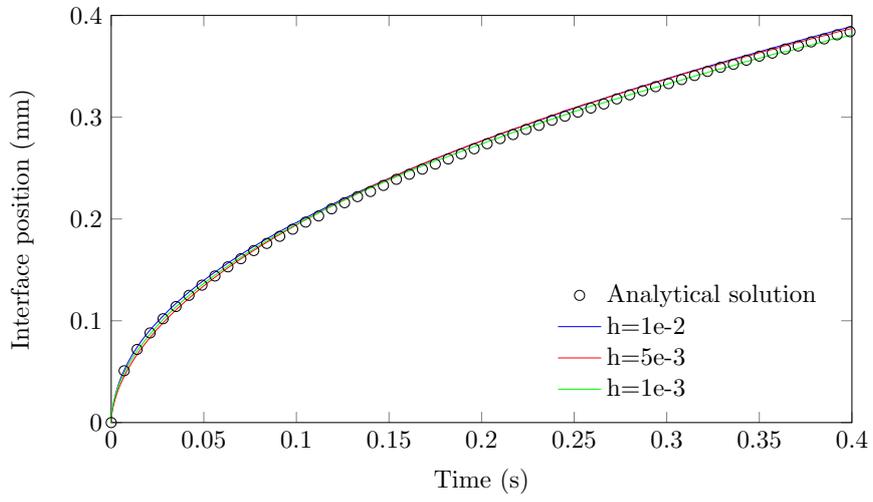


Figure 8: Evolution of the interface position for the Stefan problem.

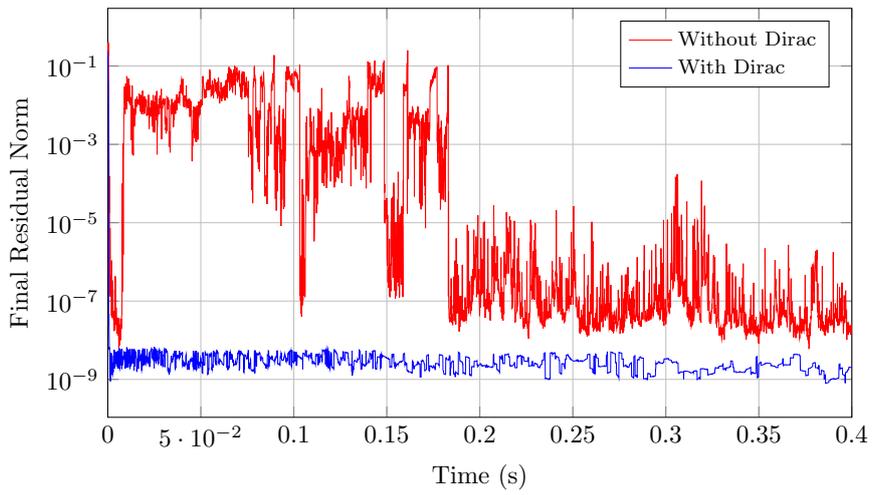


Figure 9: Final residual norm for the level set solver.

343 *4.2. 2D Film boiling*

344 This test case serves as a validation for the phase change model in several articles [1, 8, 14, 54]. Indeed,
 345 when the temperature at the wall is much larger than the saturation temperature, a persistent layer of
 346 vapor forms and remains between the wall and the water. This regime is known as film boiling. In the
 347 configuration depicted in Figure 10, a Rayleigh-Taylor instability is triggered due to the low density fluid
 348 below the high density fluid. Furthermore, the phase change will induce a growth of the film, amplifying
 349 the instability.

For a Rayleigh Taylor instability, the most unstable Taylor wavelength is, in 2D,

$$\lambda_0 = 2\pi \left(\frac{3\gamma}{(\rho_l - \rho_g)g} \right)^{1/2}, \quad (59)$$

350 where γ is the surface tension.

351 The physical parameters taken from [8] are given in Table 2. For this set of parameters, the most unstable
 352 wavelength is about $\lambda_0 \approx 0.078\text{m}$. Figure 10 shows the initial profile of the interface, defined by the following
 353 function:

$$y = 0.5 + 0.4 \cos(2\pi x / \lambda_0). \quad (60)$$

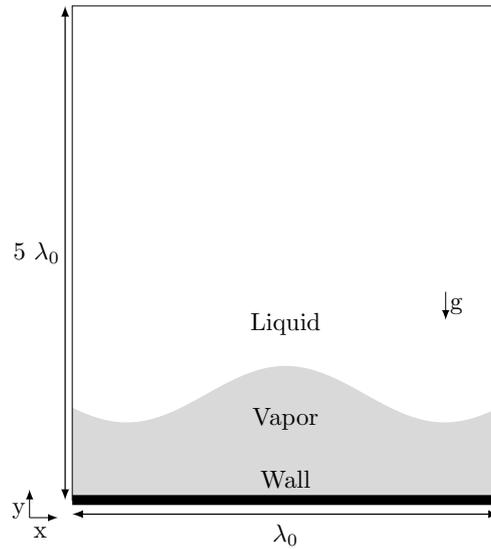


Figure 10: Setup for the 2D film boiling

354 Two cases will be studied. For the first case, the temperature at the wall is maintained constant at a
 355 temperature of 5K above the saturation temperature and for the second, 10K. To assess the accuracy of the
 356 computations, the authors in [1, 8, 54] used space-averaged Nusselt number obtained from the numerical

357 simulation and compare it to correlation found in the literature. The local Nusselt number defined as the
 358 dimensionless heat flux through the wall is

$$Nu_u = \frac{\lambda_0}{T_w - T_{\text{sat}}} \left. \frac{\partial T}{\partial y} \right|_{y=0}. \quad (61)$$

359 Regarding the correlation of the Nusselt number in the literature, we use the correlation of Berenson [55]
 360 and Klimenko [56] given by

$$Nu_B = 0.425(GrPr/Ja)^{1/4}, \quad (62)$$

361 and

$$Nu_K = 0.1691(GrPr/Ja)^{1/3} \text{ for } Ja < 0.71. \quad (63)$$

362 where the Grashof number $Gr = \rho_v(\rho_l - \rho_v)g\lambda_0/\mu_v^2$ represents the ratio of the buoyancy force over the
 363 viscous force, the Prandtl number $Pr = \mu_v c_p^v/k_v$ represents the ratio of the momentum diffusivity over the
 364 thermal diffusivity and the Jakob number $Ja = c_p^v(T_w - T_{\text{sat}})$ the ratio of sensible heat over latent heat.

Table 2: Density, dynamic viscosity, specific heat and thermal conductivity for the vapor and the water at atmospheric pressure

	ρ [kg/m ³]	μ [Pa·s]	c_p [J/(kg·K)]	k [W/(m·K)]	L_{sat} [J/kg]
Vapor	5.0	0.005	200	1.0	
Water	200	0.1	400	40	10^4

365 Figure 11 shows the evolution of the temperature field and the interface for $\Delta T = 10K$. The vapor film
 366 grows to a mushroom shape due to the Rayleigh-Taylor instability. Since we performed 2D computations,
 367 no break up occurs due to surface tension. Therefore the mushroom cap rises along the channel followed
 368 by a long and thin filament. Notice that due to the width of the channel, there is clearly an effect of the
 369 lateral confinement on the shape of the vapor mushroom. Figure 12 shows the evolution of the mesh. The
 370 mesh is adapted using the anisotropic mesh adaptation procedure presented in this paper, using only 25,000
 371 elements. Therefore the mesh is very fine at the vapor/water interface and at the bottom when the thermal
 372 gradient is the largest. The mesh remains coarse far from the interface, where the thermal gradient is null
 373 and the velocity field is still 0.

374 The evolution of the space-averaged Nusselt number is depicted in Figure 13. The comparison with the
 375 correlation of Berenson and Klimenko shows a good agreement for both cases.

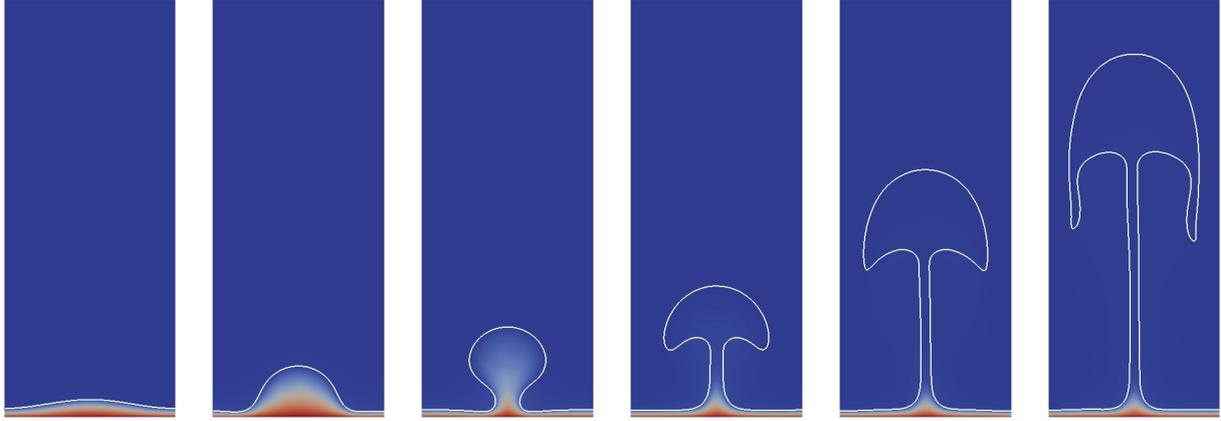


Figure 11: 2D Film boiling for $\Delta T = 10K$. Temperature field and interface location at $t=0.01, 0.2, 0.3, 0.4, 0.7$ and $1.0s$. The interface is represented by the white line.

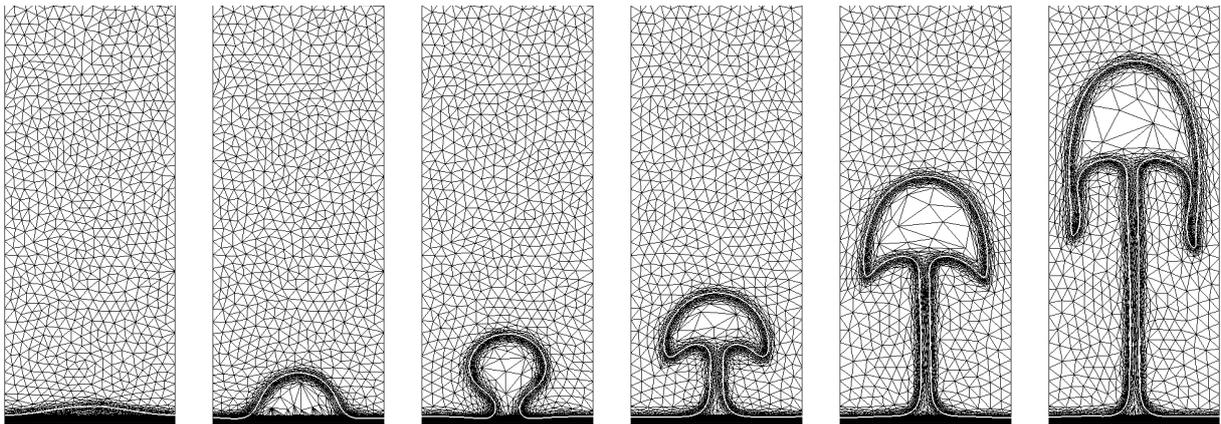


Figure 12: 2D Film boiling for $\Delta T = 10K$. Evolution of the mesh and the interface location at $t=0.01, 0.2, 0.3, 0.4, 0.7$ and $1.0s$. The interface is represented by the white line.

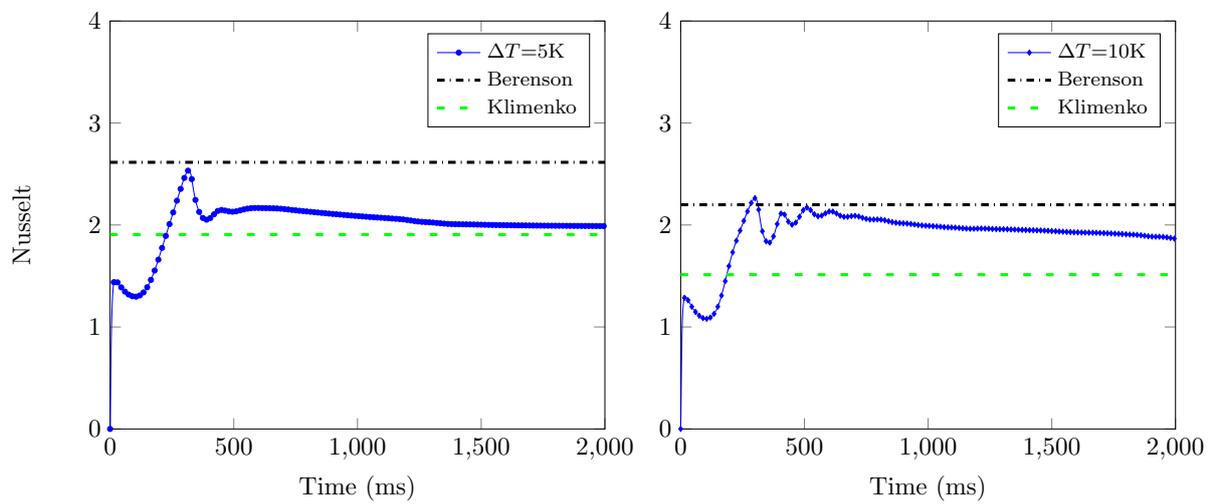


Figure 13: Evolution of the space-averaged Nusselt number for $\Delta T = 5K$ (left) and $\Delta T = 10K$ (right).

376 4.3. 3D boiling - sensitivity analysis

377 Several studies show that quench severity is dependent on different parameters: orientation, position,
378 shape and size, agitation rate, fluid viscosity, and other variables. In this section, we will analyze at least
379 the first three parameters. The objective is to show that the physical phenomena taking place are rich and
380 that the experimental investigation remains generally limited. Indeed, the latter suffers from systematic
381 revalidation of heat transfer coefficients when dealing with each parameter.

382 We consider then a water tank of dimension $L \times L \times L$ (see Figure 14), filled to three quarters. Two
383 geometries are proposed; a hot metallic cylinder of length 0.5m and diameter 0.1m with $L=1$ m and a hot
384 hollow cylinder (see Figure 18) of inner radius 3cm and outer radius of 6cm with $L=0.40$ m.

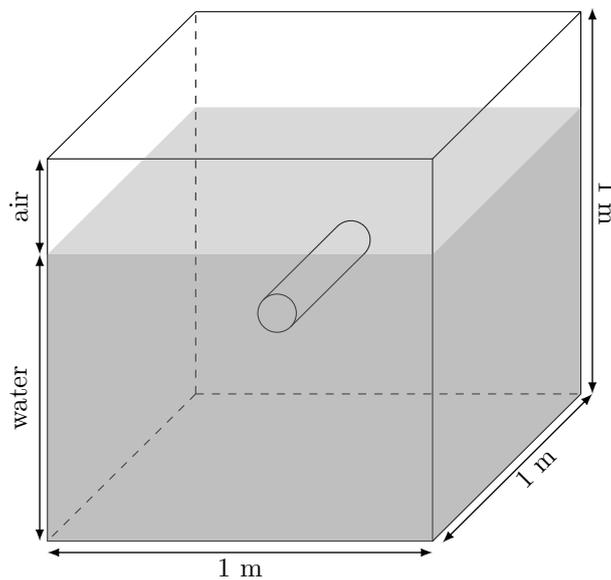


Figure 14: Set-up for the 3D film boiling

385 For the first geometry, three different configurations are considered. The cylinder is immersed horizontally
386 at mid-height (see Figure 15), at a quarter of the height of the tank (see Figure 16) or vertically (see Figure
387 17).

388 Figure 11 shows a persistent vapor film surrounds the cylinder. As expected, the position of the cylinder
389 has an important effect on the film evolution and thus on the cooling velocity of the cylinder. Furthermore,
390 such distance from the cylinder to the free surface has a direct impact on the overall flow inside the quenching
391 bath. The higher the distance, the higher the velocity of the vapor phase breaking up at the surface.

392 Whereas, when the cylinder is immersed vertically inside the water tank, the shape of the film is totally
393 different. Indeed, we can notice in Figure 12 a periodic release of the film all along the surface. Thus, the
394 distribution of the flow is more concentrated at the top of the **cylinder**. This induces lower cooling velocity
395 when compared to the previous case.

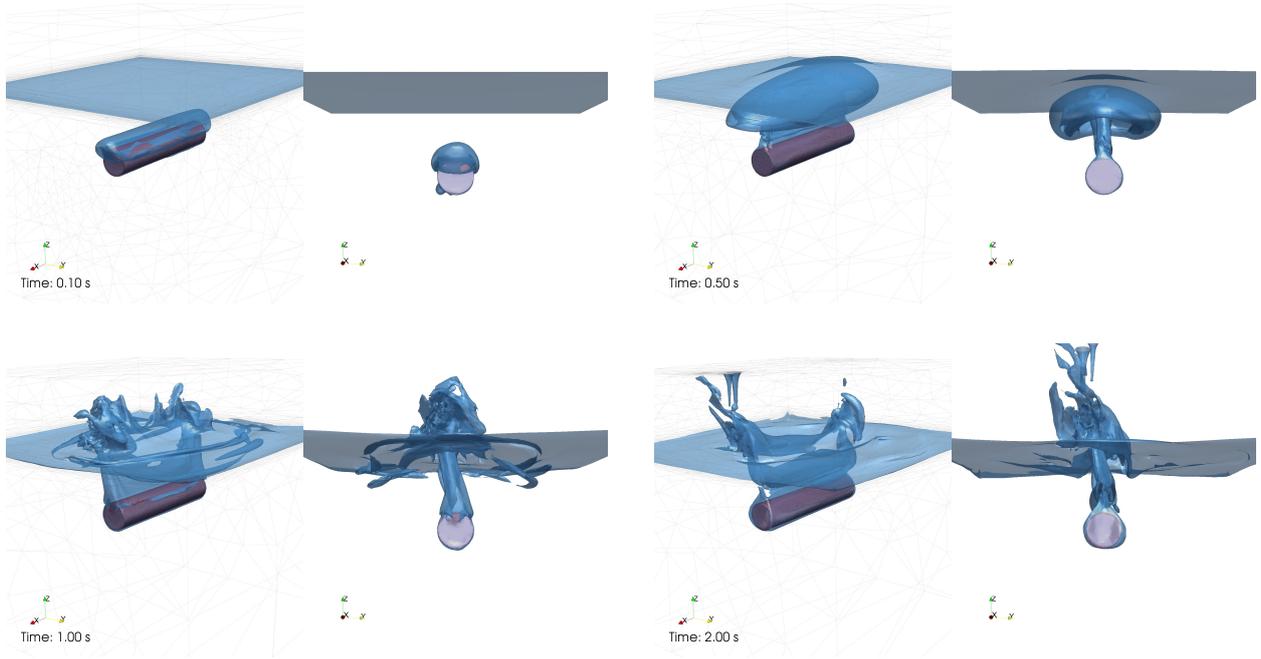


Figure 15: Cylinder at mid-height. Results at $t=0.1, 0.5, 1$ and 2 s. Perspective view (left), front view (right).

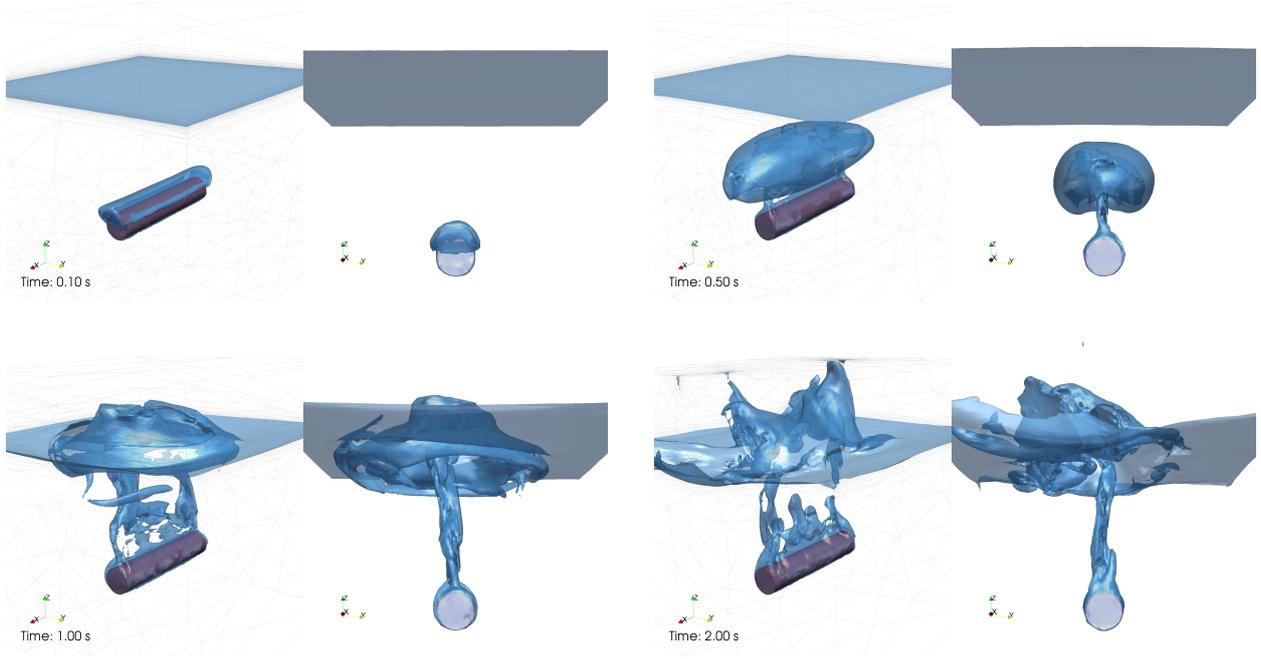


Figure 16: Cylinder at a quarter of the height. Results at $t=0.1, 0.5, 1$ and 2 s. Perspective view (left), front view (right).

396 Finally, Figure 19 shows this time that both the size of the **hollow** cylinder and its geometrical features
 397 seem to affect completely the flow. The boiling is well guided by the top surface with additional concentration

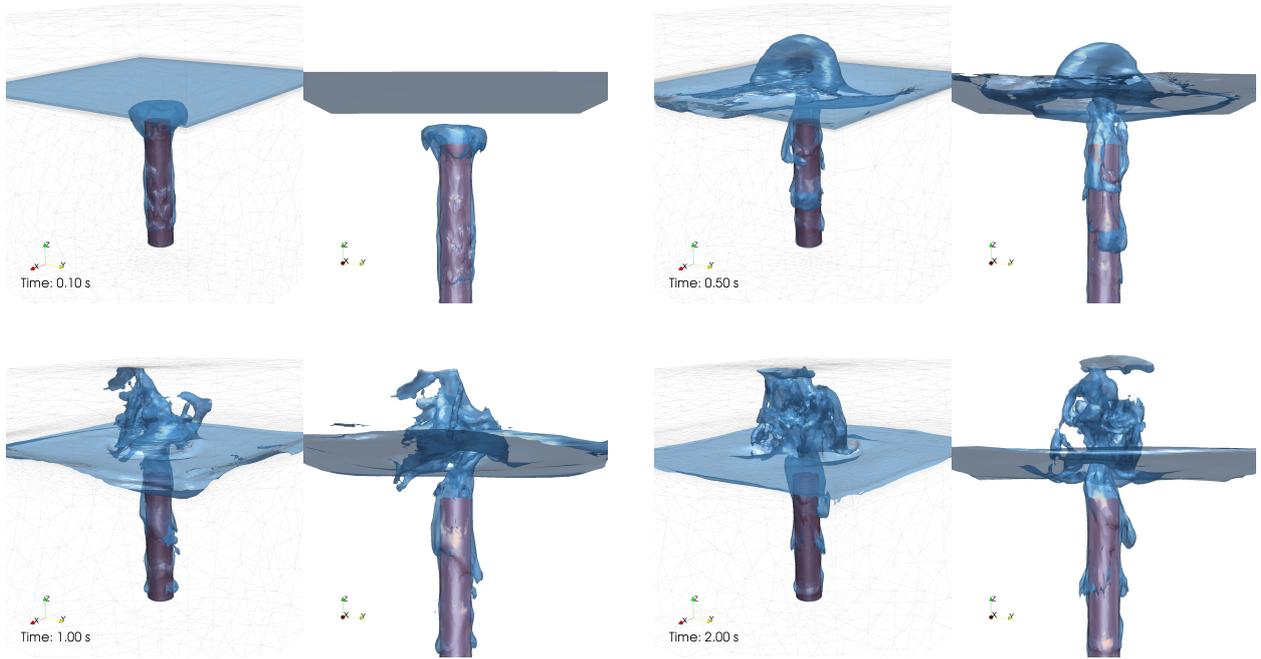


Figure 17: Vertical cylinder. Results at $t=0.1, 0.5, 1$ and 2 s. Perspective view (left), front view (right).

398 along the extremity. This again confirms the important role of this adaptive Eulerian framework to predict
 399 several liquid-vapor phase changes during boiling as well as to handle easily optimal combination of quench
 400 parameters.

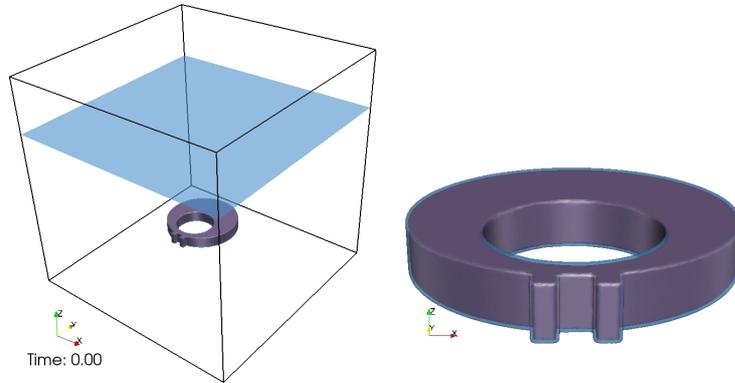


Figure 18: Hollow cylinder : Full domain with the ring (in red) and the gas-liquid interface in blue(left). Zoom on the ring (right)

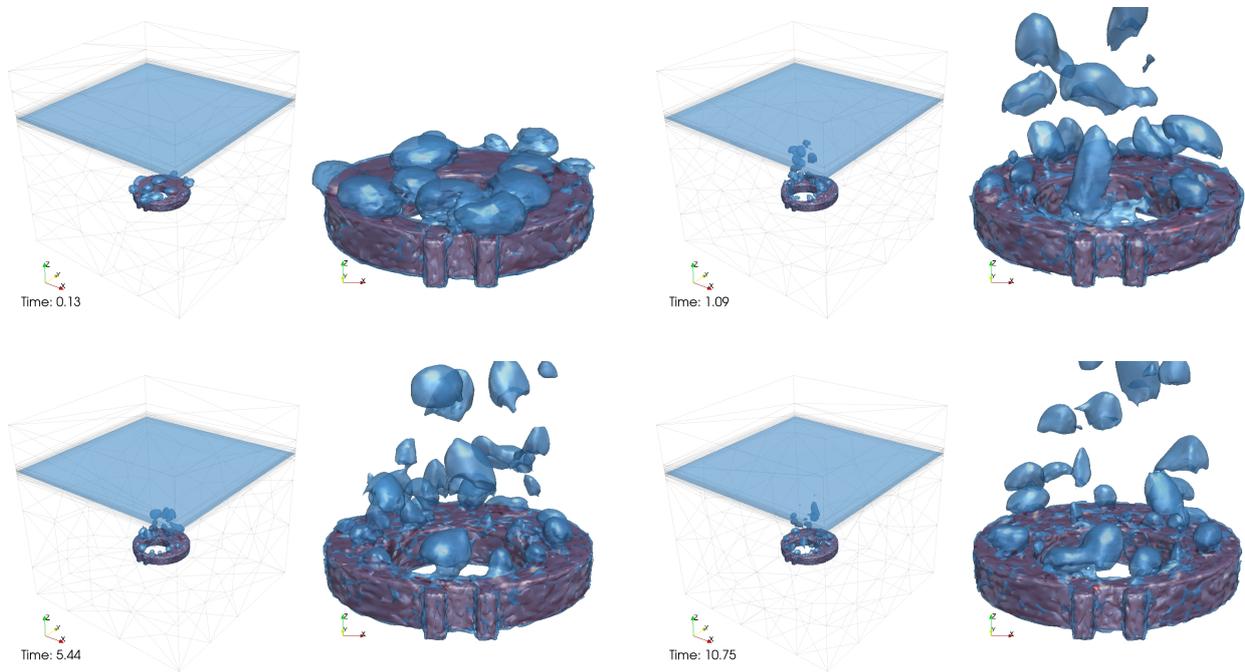


Figure 19: Hollow cylinder. Results at $t=0.13, 1.09, 5.44$ and 10.75 s. Perspective view (left), front view (right).

401 4.4. Quenching of a solid, comparison with experimental data

402 We consider a domain of size $0.60 \times 0.60 \times 0.40 \text{ m}^3$, three-quarters full of water, in which a metallic
 403 sample of dimension $0.075 \times 0.075 \times 0.0015 \text{ m}^3$ is immersed (see Figure 20). The temperature of the sample
 404 is $T_{\text{solid}} = 880^\circ\text{C}$ and the temperature of the water is $T_{\text{water}} = 25^\circ\text{C}$. A thermocouple is placed at the core of
 405 the metallic sample. From a practical point of view, due to the small thickness of the part, the temperature
 406 at the core reflects the behavior of the temperature field at the interface.

407 A free slip boundary condition is prescribed on all the walls. The motion of the vapor film is due to the
 408 buoyancy force and the surface tension force. The simulations are performed using the proposed adaptive
 409 meshing technique with 200,000 tetrahedra. The time step is set to $\Delta t = 0.002\text{s}$.

410 All the physical parameters related to each phase are presented in Table 3.

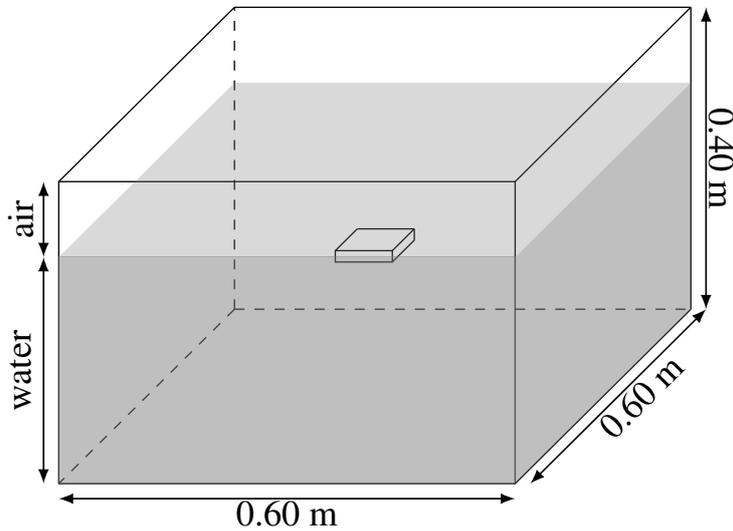


Figure 20: Set-up for the 3D industrial quenching

Table 3: Physical parameters defining the test case for the 3D industrial quenching

	ρ [kg/m ³]	μ [Pa·s]	c_p [J/(kg·K)]	k [W/(m·K)]	L_{sat} [J/kg]
Vapor	1.0	1.2×10^{-5}	2010	0.025	2.26×10^6
Water	1000	1.0×10^{-3}	4185	0.6	
Solid	8000		435	11.4	

411 The evolution of the liquid-vapor phase is depicted in Figure 21 and shows again the ability of the
 412 proposed Eulerian framework with adaptive meshing to deal with such challenging test case.

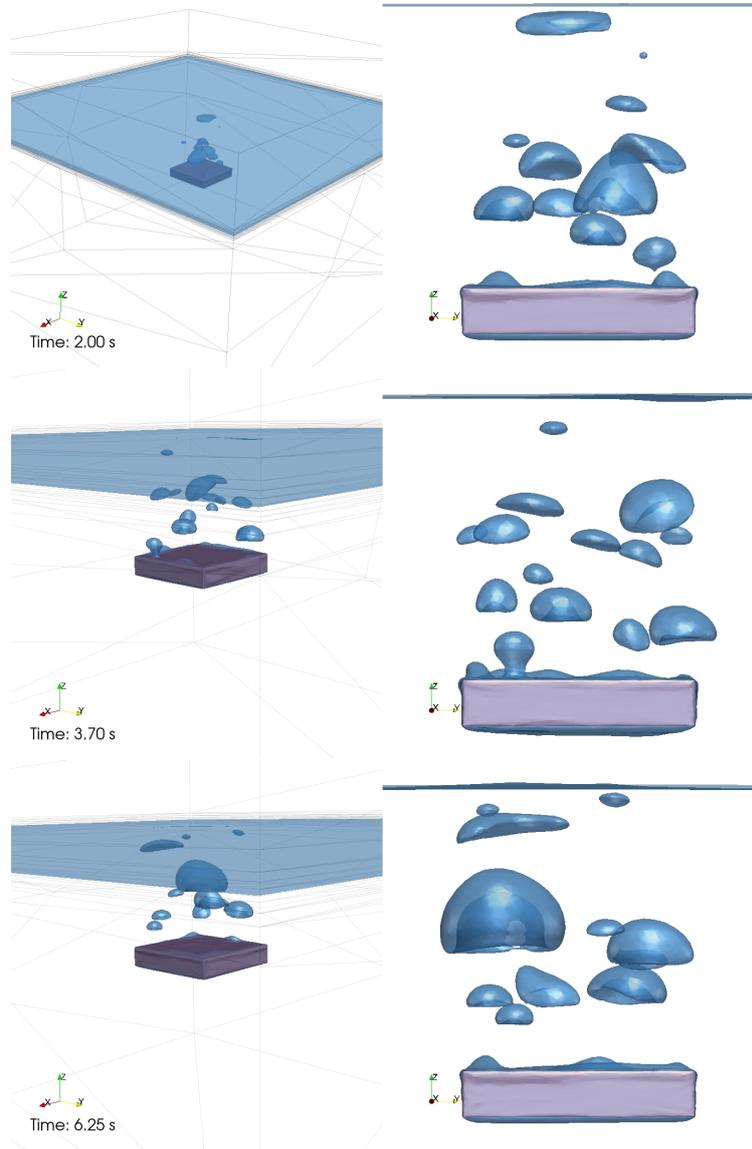


Figure 21: Industrial quenching. Results at $t=2, 3.70$ et 6.25 s. Perspective view (left), front view (right). The zero isovalue of the level set is represented in blue.

413 Figure 22 represents a clip of the quenching tank. Water is depicted in blue and the solid in red. A slice
 414 enables to visualize the adaptive mesh over time. At the top left corner, the picture shows the initial mesh,
 415 mostly concentrated around the part and the free surface. As the simulation starts and bubbles form, the
 416 mesh is adapted automatically at the level set interface. Under the constraint of a fixed number of elements,
 417 one can notice that the mesh is coarsened automatically at the bottom of the tank, where the variation of
 418 the solutions is small.

419 Finally, Figure 23 shows a good agreement for the temperature evolution between the experimental data

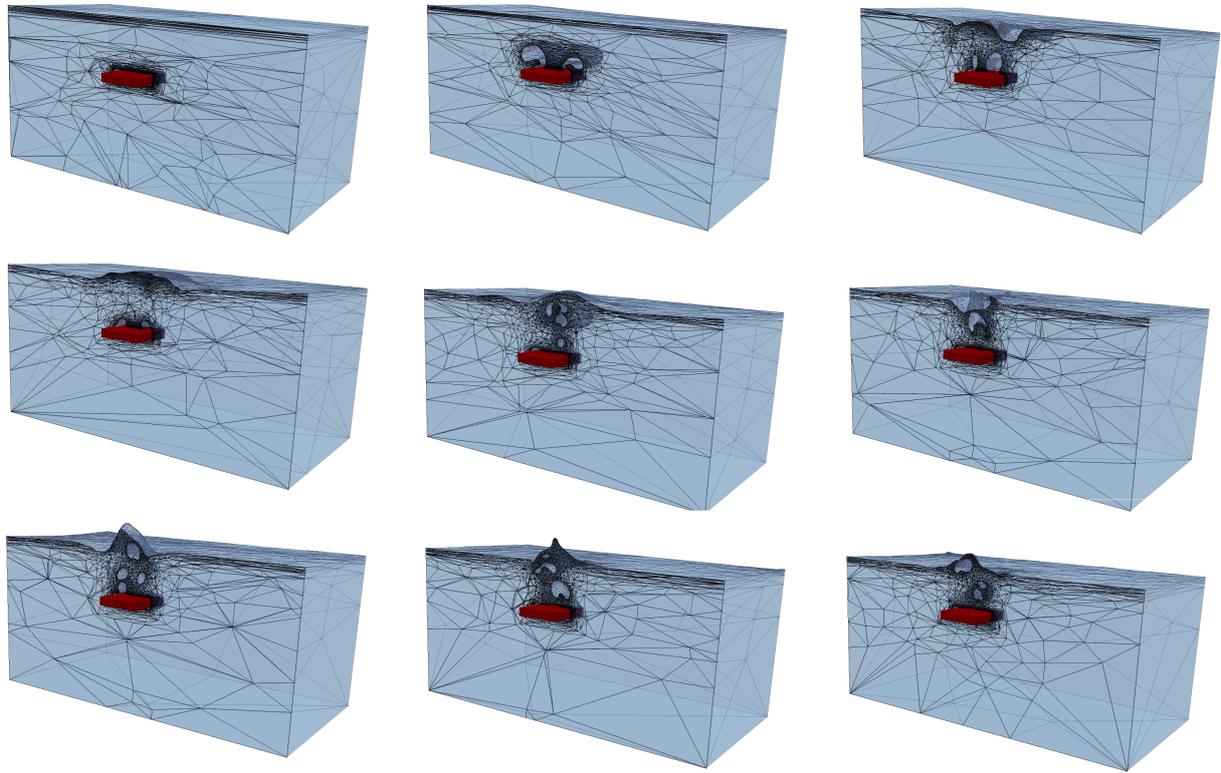


Figure 22: Industrial quenching: results at different instants. The solid is represented in red. A clip of the quenching tank, with visible adapted mesh.

420 and the numerical simulation. The strong coupling between the solid, the water and the vapor enables to
421 perform such simulation without the use of a heat transfer coefficient at the solid boundaries. The anisotropic
422 mesh adaptation enables to capture the large thermal gradient, the large jump at the interfaces as well as
423 the complexity of the flow. Simulations are performed using 100,000 to 400,000 elements.

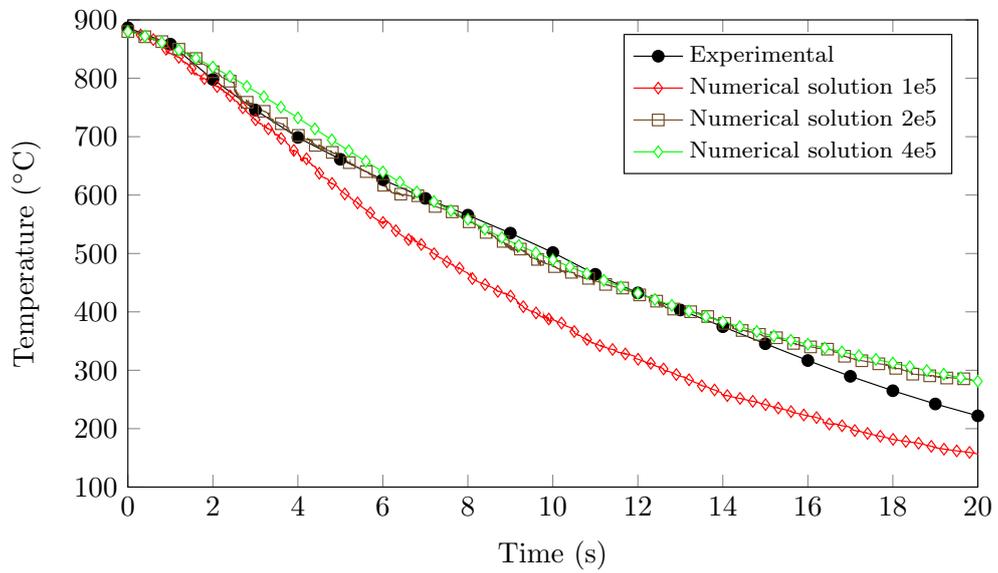


Figure 23: Evolution of the computed temperature at the core of the sample, sensitivity to the number of elements, comparison to experiments.

424 5. Conclusion

425 We proposed an adaptive Eulerian framework for the numerical simulation of the phase change and the
426 evaporation that occurs at the interface between a heated solid surrounded by cooled liquid. We showed
427 that in order to model efficiently these phenomena, the gas-liquid phase changes, the vapor formation and
428 their dynamics, and the conjugate heat transfer must be solved simultaneously. Therefore, it replaces the
429 use of classical assumptions and ad-hoc transfer coefficients. We have demonstrated the efficiency of this
430 framework by performing challenging cases in 2D and 3D with comparisons including experimental data.
431 For the first time, the direct numerical simulation of the solid-liquid-gas interaction for industrial quenching
432 can be performed accurately.

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